CH 9: Design of Permanent Joints

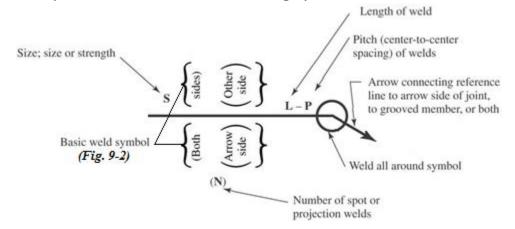
This chapter introduces permanent joining methods which include: welding, soldering, cementing, bonding, etc.

Permanent joining usually leads to <u>significant savings</u> over non-permanent joining (*because of the elimination of fasteners and holes*).

Welding Symbols

Welding symbols are used on drawings to indicate the <u>type and specifications</u> of the weldments.

• <u>Figure 9-1</u> shows the *American Welding Society* (*AWS*) standard welding symbol. The most important features of the welding symbol are illustrated below:



• The table below shows the *Basic weld symbol* for the different types of welds.

			Type o	f weld			
Bead	Fillet	Plug or slot	Groove				
			Square	V	Bevel	U	J
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- Figures 9-3, 4, 5 show some examples for the use of welding symbols.
- In general, there are two types of welds; butt welds and fillet welds.
 - <u>Fillet welds</u> are the most used type for machine elements.
 - <u>Butt welds</u> are usually used for pressure vessels.

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- Since welding is associated with a <u>significant increase in temperature</u>, there will be some <u>metallurgical</u> changes in the parent material <u>in the vicinity</u> of the weldments.
 - Thus, when checking for failure of the parent material in the vicinity of the weldments, it is recommended to use the properties of <u>Hot-Rolled</u> (HR) material even if the material is Cold-Drawn (CD).
- Also, <u>residual stresses</u> might be introduced during welding because of <u>clamping</u> or sometimes because of the <u>order</u> of welding.

Butt and Fillet Welds

- For a <u>Butt weld</u> subjected to tensile force or shear force;
- The normal stress is found as:

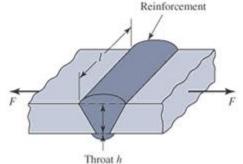
$$\sigma = \frac{F}{hl}$$

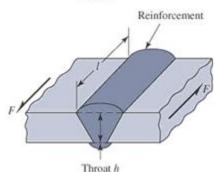
Where *h*: Throat of the weld

(Does not include the reinforcement)

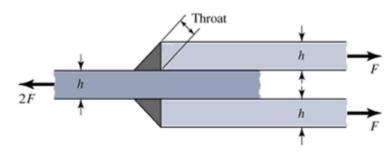
- *l*: Length of the weld
- The average shear stress is found as:







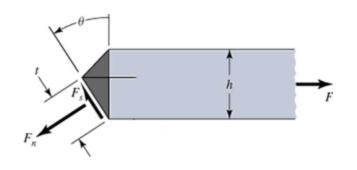
- The reinforcement causes <u>stress concentration</u> and therefore it should be <u>removed</u> by grinding or machining if the joint is subjected to <u>fatigue</u> loading.
- For a <u>Fillet weld loaded in tension;</u>
 - The forces in each weldment will have <u>two components</u>. Normal force "F_n" and Shear force "F_s" where their magnitudes change with the angle.



• From equilibrium the magnitudes are found to be:

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$$F_n = F \cos \theta$$
$$F_s = F \sin \theta$$



• From trigonometry the throat "t" is found as:

$$t = \frac{h}{\cos\theta + \sin\theta}$$

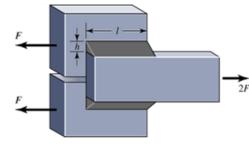
• Therefore, <u>at any angle</u> θ there are normal stress " σ " and shear stress " τ " where;

$$\sigma = \frac{F_n}{A} = \frac{F_n}{lt} \qquad \qquad \& \qquad \tau = \frac{F_s}{A} = \frac{F_s}{lt}$$

- > The values of $\sigma \& \tau$ depend on θ and using *Von Mises* stress, the <u>maximum</u> stress is found to occur at $\theta = 62.5^{\circ}$.
- However this analysis procedure is <u>too complicated</u> and the geometry of weldments is <u>not uniform</u>.
- Instead, a <u>simplified</u> (*conservative*) <u>approach</u> is used for design purposes.
 - The approach <u>ignores the normal stresses</u> and assumes that the external force is carried as <u>shear stress only</u> on the <u>smallest throat</u> area of the weldment (*the smallest value of "t"* @ $\theta = 45^{\circ}$).
 - Conservative because all the external force is assumed to cause <u>shear</u> stress knowing that the <u>shear strength</u> is almost <u>half</u> of the normal strength.
 - According to this approach, the shear stress is found as:

$$\tau = \frac{F}{lt} = \frac{F}{l(h\cos 45)} = \boxed{\frac{1.414 F}{l h}}$$

• The <u>same equation</u> is also used to calculate the shear stress when a fillet weld is <u>loaded in shear</u>.



• For a welded joint loaded with <u>eccentric</u> force there will be, in general, more than one shear stress component acting

on the weldments where there will be a <u>primary</u> shear due to the force itself and a <u>secondary</u> shear resulting from the <u>moment or torque</u> produced by the force.

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Stresses in Welded Joints in Torsion

For a welded shear joint such as the shown there will be two shear stress components in the weldments.

• <u>Primary shear</u> (due to the shear force).

$$\tau' = \frac{V}{A}$$

Where A is the <u>total throat area</u> of all the welds.

• <u>Secondary shear</u> (due to the twisting moment "the torque").

$$\tau^{\prime\prime} = \frac{Mr}{J}$$

Where: r is the distance from the <u>centroid</u> of the weld group to the <u>point of</u> <u>interest</u>.

J is the polar moment of inertia of the weld group (*based on the throat area*) about the centroid of the group.

The centroid can be located as:

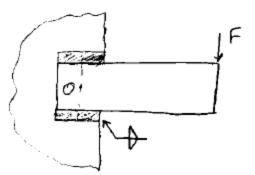
$$\bar{x} = \frac{A_1 x_1 + A_2 x_2 + A_3 x_3 + \dots}{A_1 + A_2 + A_3 + \dots}$$
$$\bar{y} = \frac{A_1 y_1 + A_2 y_2 + A_3 y_3 + \dots}{A_1 + A_2 + A_3 + \dots}$$

- To make the calculation <u>easier</u>, the polar moment of inertia for some common welding patterns is given in tables.
 - ✤ <u>Table 9 1</u> gives the location of the centroid and the <u>unit</u> polar moment of inertia " J_u " for some common weld shapes.
 - > J_u is the polar moment of inertia for a line (assuming unit width).
 - Since the minimum <u>throat width</u> for a fillet weld is (0.707 h) then J is found as:

$$J = 0.707 h J_u$$

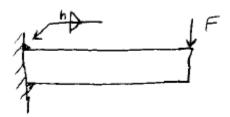
See Example 9-1 from text

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Stresses in Welded Joints in Bending

For the cantilever beam welded to a support using fillet wilds as shown, the welds will be subjected to two components of shear stress.



• <u>Primary shear</u> (*due to the shear force*).

$$\tau' = \frac{V}{A}$$

Where *A* is the *total throat area* of all the welds.

- <u>Secondary shear</u> (*due to the tensile force produced by the bending moment*). The moment will cause a <u>bending stress</u> of magnitude $\sigma = \frac{Mc}{I}$. However, according to the <u>conservative approach we use</u>, all the force acting on fillet weld is assumed to be <u>carried as shear</u> force on the throat area of the weld.
 - > Thus we can write;

$$\tau^{\prime\prime} = \frac{Mc}{I}$$

Where: *c* is the distance from the <u>neutral axis</u> of the weld group to the <u>point of</u> <u>interest</u>.

I is the moment of inertia of the weld group (*based on the throat area*) about the neutral axis of the group.

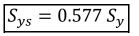
- * <u>Table 9 2</u> gives the location of the centroid and the <u>unit</u> moment of inertia " I_u " for some common weld shapes.
 - The moment of the inertia of the throat area is found as:

$$I = 0.707 h I_u$$

The Strength of Welded Joints

The electrode materials (filler material) are standardized and they are usually chosen such that they have <u>higher strength</u> than the parent materials.

- ✤ <u>Table 9 3</u> gives the minimum properties for the AWS electrode classes.
 - It should be noted that the table gives the <u>"tensile" yield strength</u> of the electrode material.
 - > The <u>"shear" yield strength</u> is found using the distortion energy theory as:



- The <u>welding code</u> includes a <u>1.6 design factor</u> for the allowable shear stress in weldments: $\tau_{all} = (0.577 S_y)/1.6$
 - Table 9 6 gives the allowable shear stress in fillet welds for the different electrode classes (with 1.6 design factor included).
- As mentioned earlier, the parent material in the vicinity of the weldment will be subjected to high temperatures and thus its <u>properties might change</u>. Therefore, if the parent material is *Cold-Dawn* it will be heat treated and thus it will have properties similar to a *Hot-Rolled* material (*in the vicinity of the weldment*).
- According to the <u>welding code</u>, the <u>allowable values of shear & normal stresses</u> in the <u>base material</u> are:

 $\tau_{all} = 0.4 S_y$ & $\sigma_{all} = 0.6 S_y$ S_y of the base material

- The procedure for evaluating the strength of welded joints is as follows:
- Find the primary shear stress due to external forces.
- Find the <u>secondary shear</u> stress due torsion *and/or* bending moments.
- Add the primary and secondary shear stress components using vector summation.
- Find the strength of <u>weldments</u>, and thus the allowable load.
- Find the strength of the <u>parent material</u> and thus the allowable load.

See Examples 9-2 & 9-4 from text

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